# Learning Control at Intermediate Reynolds Numbers

# Experiments with perching UAVs and flapping-winged flight

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• Reinforcement learning has the potential to generate high-performance <u>nonlinear</u>, adaptive control policies for complicated systems...

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Control in complicated fluid dynamics

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#### Example: Landing on a perch



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## The State-of-the-Art in Perching

Approach 1: Morphing plane





January 28th, 2007

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- Both sacrifice performance to use linear control on modified vehicles (can't compete w/ birds!)
- Can learning nonlinear control produce superior performance on existing vehicles?

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#### **Technical challenges for control**

- Dynamics quickly get too complicated for conventional control design
  - Fluid dynamics are time-varying and very nonlinear
  - CFD simulations in these regimes can take days to compute
  - Severe lack of compact (design accessible) models
- Limited control authority
  - Flow is only partially observable
  - Stalls result in intermittent losses of control authority
  - "Underactuated control" control actions have long-term consequences

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#### Learning Control for Perching

- Accurate Navier-Stokes simulations takes days to compute, but...
- Model-based Reinforcement Learning:
  - Learn approximate model of unsteady dynamics from real flight data
  - Formulate the goal of control as the long-term optimization of a scalar cost
  - **3** Offline model-based numerical optimal control
  - Online model-free optimal control ("learning")

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#### **Experiment Design**



- Glider (no propellor)
- Dihedral (passive roll stability)
- Offboard sensing and control



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## **System Identification**

- Nonlinear rigid-body vehicle model
- Linear (w/ delay) actuator model
- Real flight data
  - Very high angle-of-attack regimes
  - Surprisingly good match to theory
  - Vortex shedding



#### Lift Coefficient

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#### Drag Coefficient

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### A dynamic model



- Planar dynamics
- Aerodynamics from model
- State:  $\mathbf{x} = [x, y, \theta, \phi, \dot{x}, \dot{y}, \dot{\theta}]$
- Only actuator is the elevator angle,  $\mathbf{u}=\dot{\boldsymbol{\phi}}$

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#### **Feedback Control Design**

• Optimal feedback control formulation:

$$\begin{aligned} J^{\pi}(\mathbf{x}) &= \min \left[ (\mathbf{x} - \mathbf{x}^d)^T \mathbf{Q} (\mathbf{x} - \mathbf{x}^d), \quad J^{\pi}(\mathbf{x}') \right], \\ \mathbf{x}' &= f(\mathbf{x}, \pi(\mathbf{x}))) \end{aligned}$$

- x is the estimated state (from motion capture)
- $\pi$  is the feedback policy, commanding elevator angle as a function of state
- f is the identified system model
- **Q** is a positive definite cost matrix
- Discretize dynamics on a mesh over state space
- Optimized Dynamic Programming algorithm approximates the optimal policy, π<sup>\*</sup>(x), from model

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#### **Glider Perching**

- Enters motion capture @ 6 m/s.
- Perch is < 3.5 m away.
- Entire trajectory @ 1 second.

#### Requires Separation!



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#### Flow visualization (very preliminary)



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#### Transition to Outdoors: Gust Response

- Current experiments are in "still air"
- Outdoor flights require robustness to
  - ambient wind conditions
  - persistent flow structures (e.g., around the perch)
  - short-term "gust" disturbances
- Capture statistical environment model
- Instrument motion capture arena with known aerodynamic disturbances, obstacles, and wind
- Already performed detailed LTV controllability analysis with different actuator configurations

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#### The Intermediate Reynolds Number regime

- Reynolds number (Re) dimensionless quantity that correlates with the resulting kinematics of the fluid
  - Low Re viscousity dominates, flow is laminar
  - High Re turbulence
  - Intermediate Re complicated, but structured flow (eg, vortex shedding). Glider perching example is Re 50,000 down to Re 15,000.
- At Intermediate Re:
  - Lots of interesting control problems
  - Almost no good control solutions

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#### **Bird-scale flapping flight**



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#### **Autonomous Flapping-Winged Flight**



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#### **Motivation**

- Bird-scale flapping vehicles will not surpass the speed or efficiency of fixed-wing aircraft for steady-level flight in still air
  - Propellors produce thrust very efficiently
  - Aircraft airfoils can be highly optimized (for speed or efficiency)

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#### **Motivation**

- Bird-scale flapping vehicles will not surpass the speed or efficiency of fixed-wing aircraft for steady-level flight in still air
  - Propellors produce thrust very efficiently
  - Aircraft airfoils can be highly optimized (for speed or efficiency)
- But looking more closely...

### **Efficient flying machines**

 An albatross can fly for hours (or even days) without flapping, even migrating upwind (exploiting gradients in the shear layer)



• Butterflies migrate thousands of kilometers, carried by the wind

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### Super-maneuverability

• Peregrine falcons have been clocked at 240+ mph in dives, and have the agility to snatch moving prey



- Bats have been documented...
  - Catching prey on their wings
  - Manuevering through thick rain forests at high speeds
  - Making high speed 180 degree turns
  - ...

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• The secret:

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• The secret:

# Birds (and fish, ...) exploit unsteady aerodynamics at intermediate Re

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• Birds far surpass the performance of our best engineered systems (especially UAVs) in metrics of efficiency, acceleration, and maneuverability.

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# Birds (and fish, ...) exploit unsteady aerodynamics at intermediate Re

- A manipulation problem
  - Requires unconventional mechanical and control designs
  - Once you start thinking of bird flight as manipulating the air, it becomes harder to appreciate fixed-wing flight

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#### **Example: Efficient swimming upstream**





from George Lauder's Lab at Harvard (Liao et al, 2003)

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#### **Prospects for machine learning**

- Key observations about fluid-body interactions at intermediate Re
  - Considerable previous work in system identification permits the use of approximate models
  - Won't always be able to discretize the state space
  - Relatively compact policies (few parameters) can generate a large repetoire of behaviors

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  - Relatively compact policies (few parameters) can generate a large repetoire of behaviors
- Formal analysis of the policy gradient algorithms reveals:
  - Performance (via SNR) degrades with the number of control parameters
  - Performance is (locally) invariant to the complexity of the plant dynamics

#### The Heaving Foil

#### work with Jun Zhang (NYU Courant)

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#### The Heaving Foil



- Rigid, symmetric wing
- Driven vertically
- Free to rotate horizontally

[Vandenberghe et al., 2006]

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#### Symmetry breaking leads to forward flight



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#### **Flow visualization**



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### Effect of flapping frequency



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#### The control problem

- Previous work only used sinusoidal trajectories
- Optimize stroke form to maximize the "efficiency" of forward flight
  - Add vertical load cell (measures  $F_z(t)$ )
  - Dimensionless cost of transport:

$$c_{mt} = \frac{\int_{T} |F_z(t)\dot{z}(t)|dt}{mg \int_{T} \dot{x}(t)dt}$$

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• Fortunately

$$\min c_{mt} = \min \frac{\int_{T} |F_z(t)\dot{z}(t)|dt}{\int_{T} \dot{x}(t)dt}$$

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#### **Prospects for optimization**

• CFD model[Alben and Shelley, 2005]



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#### **Prospects for optimization**

• CFD model[Alben and Shelley, 2005]



• Takes approximately 36 hours to simulate 30 flaps

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#### **Experimental Optimization**

- Can we perform the optimization directly in the fluid?
- Direct policy search
  - Needs to be robust to noisy evaluations
  - Minimize number of trials required



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#### **Optimized policy gradient**

• The basic algorithm (weight perturbation):

- Perturb the control parameters, **p** by some amount **z** from  $N(0, \sigma)$
- Perform the update:

$$\Delta \mathbf{p} = -\eta (c_{mt}(\mathbf{p} + \mathbf{z}) - c_{mt}(\mathbf{p}))\mathbf{z}$$

Strong performance guarantees

$$E[\Delta \mathbf{p}] \propto -\frac{\partial c_{mt}}{\partial \mathbf{p}}$$

- Poor performance (requires many trials)
- SNR optimized policy gradient [Roberts and Tedrake, 2009]

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#### Learning results



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#### Learning results (cont.)



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### A dynamic explanation

- Forward speed is linear in flapping frequency
  - from experiments
  - statement about average speed
- Drag forces quadratic in speed ( $F \propto 
  ho SV^2$ )
- Triangle wave obtains highest average speed w/ minimal drag

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#### Implications

- Enabling tool for experimental fluid dynamicists
- Suggests that motor learning algorithms could produce efficient control solutions in fluids
- Suggests that we can use this to control robotic birds
- Exciting prospects for online learning in changing environments

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#### Summary



- Nonlinear, underactuated control w/ imperfect models via machine learning control (birds don't solve Navier-Stokes)
- Allows our machines to exploit unsteady flow effects
- Soon, robotic birds will:
  - Fly efficiently and autonomously
  - Outperform fixed-wing vehicles in maneuverability

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